

# Piper Aircraft Corporation

1983

Specifications/Performance  
Standard Equipment

## Saratoga/Turbo Saratoga

PA 32-301/301T





**Specifications**

	<b>Saratoga</b>	<b>Turbo Saratoga</b>
<b>ENGINE</b>		
Manufacturer	Lycoming	Lycoming
Model	IO-540-K1G5	TIO-540-S1AD
Rating (hp @ rpm) (Take-Off)	300 @ 2700	300 @ 2700
Rating (hp @ rpm) Continuous - 2 - Blade Prop	294 @ 2600	294 @ 2575
Rating (hp @ rpm) Continuous - 3 - Blade Prop	300 @ 2700	300 @ 2700
Number of Cylinders	6	6
Recommended TBO (hrs.)	2000	1800
<b>PROPELLER</b>		
Manufacturer	Hartzell	Hartzell
Number of Blades	2	2
Type	Constant Speed	Constant Speed
Diameter (in./cm.)	80/203	80/203
<b>WEIGHTS</b>		
Ramp Weight (lbs./kg)	3615/1639	3617/1640
Gross Weight (lbs./kg)	3600/1633	3600/1633
Standard Empty Weight (lbs./kg) (Includes unusable fuel, full oil and full operating fluids) (lbs./kg)	1935/878	1998/906
Standard Useful Load (lbs./kg)	1680/762	1619/734
Zero Fuel Weight (lbs./kg)	3600/1633	3600/1633
<b>WING AREA AND LOADINGS</b>		
Wing Area (ft. <sup>2</sup> /m <sup>2</sup> )	178.3/16.6	178.3/16.6
Wing Loading (lbs./ft. <sup>2</sup> )/(kg/m <sup>2</sup> )	20.2/98.7	20.2/98.7
Power Loading (lbs./hp)/(kg/hp)	12.0/5.5	12.0/5.5
<b>DIMENSIONS</b>		
Wing Span (ft./m)	36.2/11.0	36.2/11.0
Length (ft./m)	27.7/8.4	28.2/8.6
Height (ft./m)	8.2/2.5	8.2/2.5
Cabin Length (Instrument panel to rear bulkhead) (in./cm.)	124.25/315.6	124.25/315.6
Cabin Width (in./cm)	49/124.5	49/124.5
Cabin Height (in./cm)	42/106.7	42/106.7
Headroom (Seat to Ceiling)		
Front Seats (in./cm.)	38/96.5	38/96.5
Middle Seats (in./cm)	35.75/90.8	35.75/90.8
Rear Seats (in./cm)	34.5/87.6	34.5/87.6
Forward Baggage Door Size (in./cm)	16/40.6 x 22/55.9	16/40.6 x 22/55.9
Aft Baggage/Utility Door Size (in./cm)	20.5/52.1 x 26/66.0	20.5/52.1 x 26/66.0
Forward Cabin Door Size (in./cm)	35/88.9 x 36/91.4	35/88.9 x 36/91.4
Aft Cabin Door Size (in./cm)	28.5/72.4 x 28/71.1	28.5/72.4 x 28/71.1
Wheel Base (ft./m)	7.75/2.4	7.75/2.4
Wheel Tread (ft./m)	10.6/3.2	10.6/3.2
<b>FUEL CAPACITY</b>		
Total Capacity (gal./L)	107/405	107/405
Usable Fuel (gal./L)	102/386	102/386
<b>OIL CAPACITY (qts./L)</b>		
	12/11.35	12/11.35
<b>BAGGAGE</b>		
Volume (ft. <sup>3</sup> /m <sup>3</sup> )		
Forward Compartment	8/.23	8/.23
Aft Compartment	17.3/.49	17.3/.49
Capacity (lbs./kg)		
Forward Compartment	100/45.4	100/45.4
Aft Compartment	100/45.4	100/45.4



# Performance

	Saratoga		Turbo Saratoga	
<b>MAXIMUM SPEED</b> (kts./kmh) (TAS at Gross Weight)	152/282		178/329 (1) 182/336 (2)	
<b>CRUISING SPEEDS</b> (TAS at Gross Weight) Optimum Altitude	Best Power	Best Economy	Best Power	Best Economy
75% power (kts./kmh)	150/278	148/274	165/306	160/296
65% power (kts./kmh)	146/270	144/266	154/285	148/274
55% power (kts./kmh)	136/252	133/246	139/258	132/245
<b>CRUISE RANGE</b> (Cruising range includes 45 minute fuel reserve at maximum range power plus allowance for fuel used during taxi, take-off, climb at MCP, cruise at optimum altitude and stated mixture plus descent)				
	Best Power	Best Economy	Best Power	Best Economy
75% power (nm/km)	745/1380	823/1524	684/1267	780/1445
65% power (nm/km)	805/1491	911/1687	730/1353	845/1566
55% power (nm/km)	849/1572	960/1778	772/1431	860/1594
<b>FUEL CONSUMPTION</b>	Best Power	Best Economy	Best Power	Best Economy
75% power (gph/Lph)	18.0/68	16.0/61	19.9/75.2	16.5/62.5
65% power (gph/Lph)	16.0/61	13.8/52	17.2/65.2	14.0/52.9
55% power (gph/Lph)	14.0/53	11.9/45	14.6/55.3	12.3/46.6
<b>RATE OF CLIMB</b> (At Sea Level and Gross Wt.) Full Throttle (fpm/MPM)		990/302		1075/328
<b>STALL SPEED</b>	IAS	CAS	IAS	CAS
Flaps Down Full 40° (kts./kmh)	58/107	60/111	58/107	60/111
Flaps Up (kts./kmh)	62/114	66/122	62/115	67/124
<b>SERVICE CEILING</b> (100 fpm) (ft./m)		16,000/4877		*20,000+/6096+
<b>TAKE-OFF DISTANCE</b>	2-Blade	3-Blade	2-Blade	3-Blade
(Sea Level, zero wind, standard temperature)				
Ground Run (ft./m)	1183/361	1013/309	1110/338	960/293
Total over 50 ft. Obstacle (ft./m)	1759/536	1573/479	1590/485	1420/433
<b>LANDING DISTANCE</b>	Std. Brakes	H.D. Brakes	Std. Brakes	H.D. Brakes
(Sea Level, zero wind, standard temperature)				
Ground Roll (ft./m)	732/223	650/198	732/223	650/198
Total over 50 ft. Obstacle (ft./m)	1612/491	1530/466	1725/526	1640/500

\*20,000 feet is maximum approved altitude for Turbo Saratoga

(1) 2-Blade Prop.  
(2) 3-Blade Prop.

## Standard Equipment

### POWER PLANT AND PROPELLER

Engine — Lycoming 300 hp at 2700 rpm, direct drive, six cylinders, fuel injection, dual ignition with shielding (1)  
Lycoming 300 hp at 2700 rpm, turbocharged, fuel injection, six cylinders, direct drive, dual ignition with shielding (2)  
Engine Mounts — Dynafocal with internal dampers  
Propeller — Hartzell, 2 blade, constant speed  
Propeller Spinner  
Geared Starter, 12 volt  
Air Filter, dry type  
Oil Filter, full flow  
Oil Cooler with thermostatic control — two (1), one (2)  
Engine machined for vacuum pump (no drive installed)  
Muffler and Exhaust System  
Alternate Air with manual control  
Oil Quick Drain

### FLIGHT INSTRUMENTS AND INDICATORS

Piper Airspeed Indicator  
Magnetic Compass  
Sensitive Altimeter (in. and mb.)  
Ammeter  
Annunciator Panel with push to test  
Alternator inoperative  
Oil pressure low  
Gyro vacuum low (with pump installed)  
Overboost (2)  
Tachometer, recording  
Fuel Quantity Gauges, two  
Fuel Pressure Gauge (1)  
Manifold/Fuel Flow Gauge  
Oil Temperature Gauge  
Oil Pressure Gauge  
Cylinder Head Temperature Gauge (2)  
Exhaust Gas Temperature Gauges (EGT) (2)  
Stabilator Trim Position Indicator  
Rudder Trim Position Indicator



# Standard Equipment Continued

## COCKPIT, FLIGHT AND GROUND CONTROLS

Flight Primary — dual with rams horn type wheels  
Provisions for elevator trim and mike button  
Flight Trim — lower floor  
Rudder  
Stabilator  
Engine Controls — pedestal  
Throttle  
Propeller  
Mixture, with lock  
Alternate air  
Engine Controls Friction Lock  
Dual Flight Controls  
Stall Warning Horn  
Cabin Heater and Defroster Controls  
Steerable Nosewheel  
Brakes  
Pilot's toe brakes  
Copilot's toe brakes  
Parking brake  
Landing Gear, fixed  
Landing Gear Strut Fairings  
Wing Flaps, four positions (0°, 10°, 25° and 40°)  
Wing Flap Control, manual  
Fuel Control Selectors with OFF/LEFT/RIGHT  
Cabin Fresh Air Vent Controls, six — individually controlled, overhead

## ELECTRICAL PROVISIONS

14 volt, Alternator — 60 amp  
12 volt, 25 amp Hour Battery  
Master Switch Relay  
Voltage Regulator  
Overvoltage Relay  
Resettable-type Circuit Breakers  
Basic Wiring  
Rocker Switches, conveniently grouped on pilot's 'waterfall' panel  
Ammeter, shunted type  
Provision for External Power Supply Receptacle  
Circuit Breaker Panel

## AVIONICS PROVISIONS

Circuit Breaker Panel  
Provisions for Automatic Locator Beacon  
Wide selection of Optional Avionics available

## FUEL SYSTEM

Four Fuel Tanks with 107 gallons total capacity, 102 gallons usable, mechanically interconnected per side  
Engine Driven Fuel Pump  
Electric Auxiliary Fuel Pump  
Fuel Quantity Sight Gauge, inboard tanks — two  
Fuel Tank Quick Drains, four  
Fuel System Central Drain  
Fuel Sampling Bottle

## LIGHTING PROVISIONS

Large selection of optional internal and external lighting has been made available

## CABIN COMFORT SYSTEM

Heater Muff provides heat to cabin area regulated by control on copilot instrument panel with six floor heat outlets  
Heat Regulating Levers — located at each passenger seat  
Windshield Defrosters  
Cabin Air Exhaust Vents, two  
Cabin Fresh Air Vents, six — individually controlled, overhead and four — individually controlled, floor mounted

## EXTERNAL FEATURES

Three Tone Exterior Paint Design, in a wide choice of color combinations  
12" registration number  
Cabin Doors, two — right forward and left rear  
Main Wheels, 6.00 x 6 with 6.00 x 6 tires with tubes, 8 ply rating  
Nose Wheel, 5.00 x 5 with 5.00 x 5 tire with tube, 6 ply rating  
Aircraft Brakes  
Stowable Towbar  
Tie Down Rings, three  
Jack Pads  
Meets FAR Part 36 noise requirements  
Cabin and Rear Cabin Door Locks with keys  
Nose Baggage Door with flush lock with key  
Utility Door, rear baggage access and cargo loading

## COCKPIT AND CABIN APPOINTMENTS AND PROVISIONS

Choice of five interior colors, which includes: fabric and vinyl seats, vinyl side panels, wall to wall carpeting and headliner  
Upholstery — all seats of Scotchguarded, nylon fabrics with side paneling of 'soft hand' vinyls  
Pilot/Copilot Seats — fabric and vinyl with magazine storage pockets on back of each seat — seats adjust fore and aft and reclining, with inertia reel shoulder and safety belts  
Armrests — crew area, two  
Ash Trays — crew area, two  
Map Pockets, two  
Storm Window, pilot  
Glove Compartment with rollaway door  
Scuff Pads, pilot and copilot  
Two Piece Windshield  
Sun Visors, two  
Door Closing Straps, two (fore and aft cabin doors)  
Four Rear Passenger Seats — fabric and vinyl with reclining feature, seat belts, side panel ash trays — incorporates quick release feature for easy removal  
Provisions for Vertically Adjustable Pilot and Copilot Seats  
Soundproofing  
Ignition Lock with key  
Locks with keys, fore and aft cabin and nose baggage  
Luggage Compartments with security straps  
Rear cabin and hat shelf area, 17.3 cu. ft. — 100 lbs.  
Fuselage nose, 8.0 cu. ft. — 100 lbs.

Power Setting Table on Sun Visor  
Compass Card  
Compass Card Holder  
Weight and Balance Plotter  
Takeoff/Landing Check Lists on Sun Visor  
Pilot's Operating Handbook  
Aircraft Log Book  
Engine Log Book  
Certificate of Airworthiness

## PRODUCT SUPPORT

Piper Warranty Form  
Piper Service Center Directory  
Inspection Forms

(1) SARATOGA  
(2) TURBO SARATOGA

The performance information is based on an airplane flown at gross weight under standard sea level atmospheric conditions except as noted and based on the latest data available at the time of publication approval. Take-off and landing performance is optimum. Actual performance depends on pilot techniques, operating surfaces and other factors. It is the responsibility of the pilot to determine that all operations are conducted within approved limits of design gross weight, center of gravity, and in accordance with the FAA-approved Airplane Flight Manual which is the only official source of operating parameters and performance information.

In accordance with GAMA format, range provides for taxi, take-off, climb at MCP, cruise at stated mixture and descent with 45-minute reserve at maximum range power. Empty weight includes unusable fuel, full operating fluids and full oil.

Piper Aircraft Corporation reserves the right to make changes in specifications, materials, equipment or prices at any time without prior notice or to discontinue models as required.

Your Piper Dealer has listings of a wide variety of optional equipment and avionics. Items most frequently chosen by owners are packaged for factory installation at substantial price savings.



**PIPER AIRCRAFT CORPORATION**  
LOCK HAVEN, PENNSYLVANIA 17745

a BANGOR PUNTA Company

MEMBER OF GAMA